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The solutions to the homework problems will be provided. It is strongly recommended that YOU do all the homework problems. Homework: Set 1 [Questions Solutions] Set 2 [Questions Solutions] Set 3 [Questions Solutions] Set 4 [Questions Solutions] Set 5 [Questions Solutions] Set 6 [Questions Solutions] Set 7 [Questions Solutions]

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From Foundations of Aerodynamics by Kuethe and Chow, Problem 6.5.1. Show transcribed image text A wing with an elliptical planform is flying through sea-level air at a speed of 45 m/s. The wing loading  $W/S = 1000 \text{ N/m}^2$ . The wing is untwisted and has the same section from root to tip. The lift curve slope of the section  $m_0$  is 5.7.

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1. Stallings, R. L., Jr., "Low Aspect Ratio Wings at high Angles of Attack." Tactical Missile Aerodynamics, edited by M. J. Hemsch and J. N. Nielsen, Progress in ...

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4.11 Numerical Solution of Flow Past Two-Dimensional Symmetric Bodies III . 4.12 Aerodynamic Interference: Method of Images 115 . 4.13 The Method of Source Panels | 19 . Chapter : 5 . Aerodynamic Characteristics of Airfoils 132 : 5.1 Introduction 132 . 5.2 The Vortex Sheet 134 . 5.3 The Vortex Sheet in Thin-Airfoil Theory 136

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